



## 2 MICRO EJECTOR

### 2.1 Supersonic Primary Flow

In order to have large entrainment ratio, the ejector should be designed in such a way that the flow velocity of the primary flow becomes very large. In the present study, an axisymmetric convergent-divergent nozzle having a small throat diameter is used to produce a supersonic primary flow.

Figures 2 and 3 show the schematic and SEM images of the micro nozzle. The throat diameter of the nozzle is designed to be  $42\ \mu\text{m}$  and fabricated by electro-discharge machining of stainless steel. The exit diameter of the nozzle is designed to be  $60\ \mu\text{m}$ , so that the supersonic nozzle has a moderate expansion angle of  $18^\circ$ . While it is commonly known that a large expansion angle will lead to serious flow separation, a small expansion angle also has a problem of degraded expansion due to a thick boundary layer. The Knudsen number at the throat is around 0.002, so that the rarefied gas effect can be neglected.

If compared with MEMS ejectors having a flat chamber [2, 3], the ejector with the present nozzle has a truly 3-D shape, which can be made without complicated fabrication process. In addition, by utilizing an axisymmetric structure, the present design minimizes the viscous loss at the wall. However, the wall surface of the present prototype nozzle is somewhat rough, and the cross-sectional shape is not a perfect circle. This is partially because electro-discharge machining from the both inlet and outlet sides is needed.

Figures 4 and 5 respectively show the experimental results of the primary volume flow rate with increasing gauge total pressure for butane and air as the primary flow source. It is shown in quasi 1-D analysis that the primary flow can reach a supersonic state when the total pressure is larger than 0.071 MPa for butane, and 0.090 MPa for air. At low total pressure, the experimental data are somewhat lower than the results of the quasi 1-D calculation probably due to the viscous effect. However, both experimental data asymptotically approach the calculation results, when the effective throat diameter is estimated to be  $43.15\ \mu\text{m}$ . The throat Reynolds number in the present experimental condition for supersonic flow is around 3600.

### 2.2 Large-entrainment-ratio Micro Ejector

The experimental setup for ejector is shown in Fig. 6. Firstly, experiment has been conducted when butane is supplied from a gas cylinder as the primary flow source.

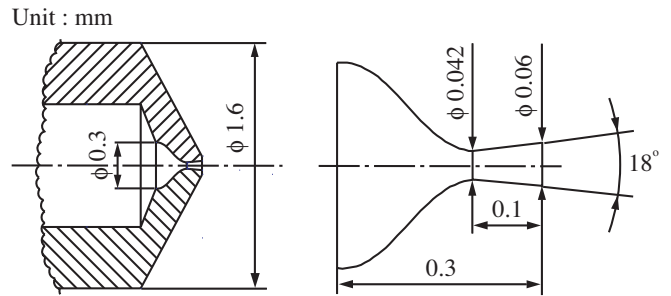


Figure 2. Schematic drawing of the micro nozzle.

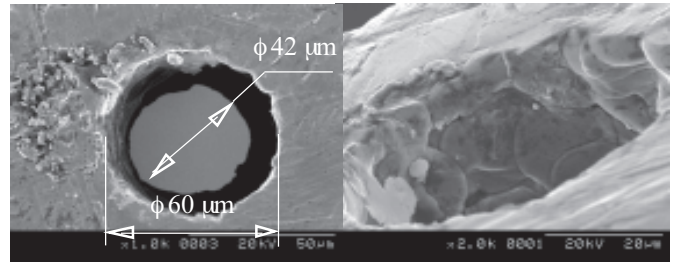


Figure 3. SEM images of the nozzle tip; a top view (left) and a magnified view (right).

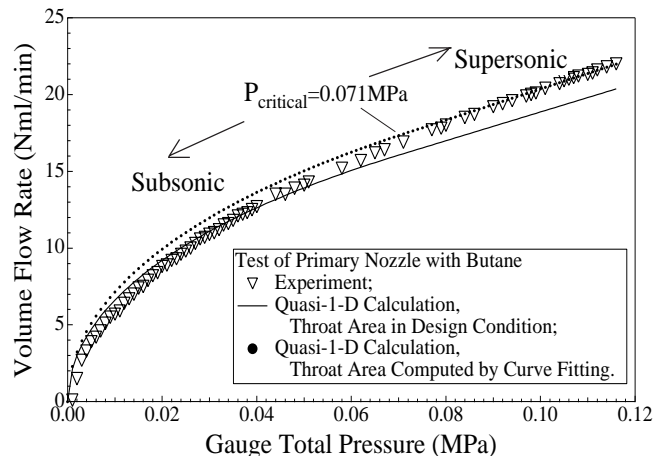


Figure 4. Effect of total pressure on the butane flow rate produced by the micro nozzle.

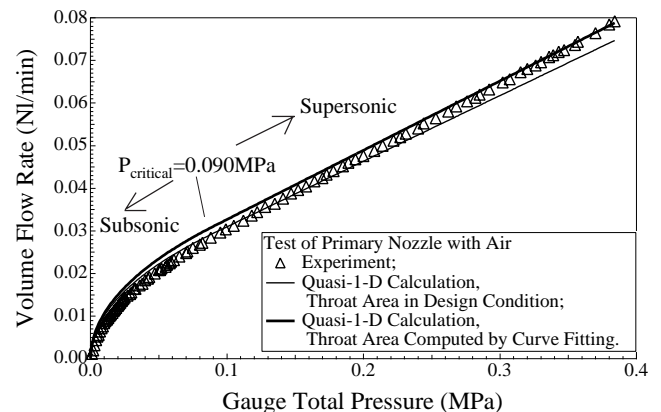


Figure 5. Effect of total pressure on the air flow rate produced by the micro nozzle.

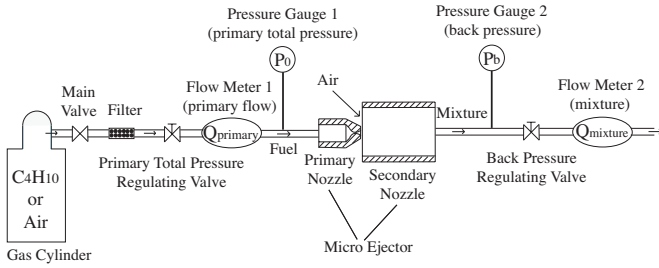


Figure 6. Experiment setup for micro ejector.

Table 1. Area ratio of micro ejectors. Unit: mm

Exit diameter of primary nozzle	0.060						
Inner diameter of secondary nozzle	1.00	1.10	1.70	2.00	2.45	3.18	4.20
Area ratio	277	335	802	1110	1666	3116	4899

Ambient air is sucked in from the inlet of the secondary nozzle. The primary total pressure and back pressure are measured by two pressure gauges. The primary flow rate and the flow rate of air-butane mixture are respectively measured by a thermal mass flowmeter and a soap film flowmeter having a low pressure drop. The area ratio is defined as the ratio of the cross-sectional area of the secondary nozzle to the exit area of the primary nozzle, and is changed by using straight circular tubes with different inner diameters (Table 1). The primary total pressure and back pressure are regulated by two valves.

Figures 7 and 8 show the effect of back pressure on the volume flow rate ratio, defined as the ratio of the secondary flow rate to the primary flow rate. The flow rate ratio reaches a maximum of 43 at a back pressure of 11.6 Pa, but the ratio rapidly decreases with increasing back pressure. The ejector with a small area ratio produces a higher volume flow rate ratio for larger back pressure as shown in Fig. 7, but it produces a lower volume flow rate ratio for small back pressure. On the other hand, the ejector with a large area ratio produces a higher volume flow rate ratio for smaller back pressure, but it produces a lower volume flow rate ratio for larger back pressure. It is shown in Fig. 7 that the increase in total pressure leads to a better performance.

CFD (Fluent 6) simulations of compressible flow in the ejector is conducted in a 2-D axisymmetric computational domain. Detailed dimensions of the domain are taken from the actual ejector. Grid dependency is examined systematically, and finally a grid system of 230\*120 is adopted. Spalart-Allmaras one-equation turbulent model [4] and species transport model are employed. The gauge total pressure of butane at the inlet of primary nozzle and gauge static pressure

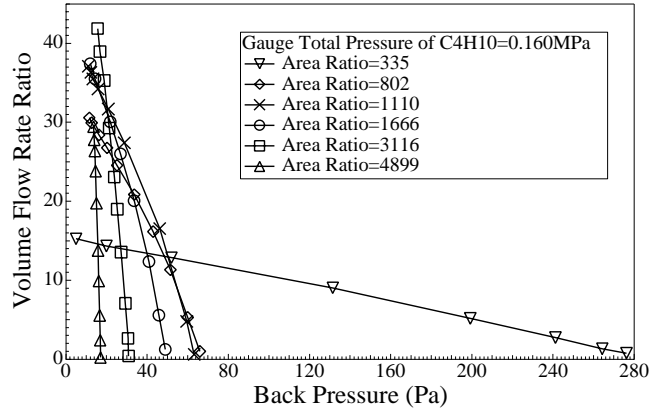


Figure 7. Effect of area ratio.

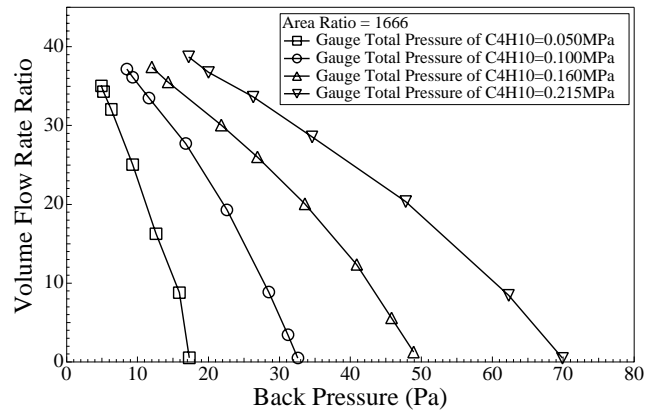


Figure 8. Effect of primary total pressure. (Butane)

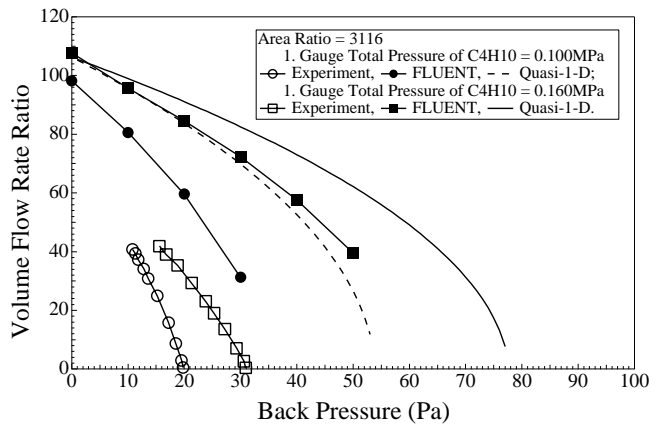


Figure 9. Comparison of volume flow rate ratio between CFD results and experimental data.

of air at the outlet are given as boundary conditions. Air at the inlet of the secondary nozzle is set as the atmospheric pressure. The wall temperature is kept at 300 K.

Figure 9 shows comparison of volume flow rate ratio between the CFD results and the experimental data. Although their trends are qualitatively similar, the present experimental data are much smaller than the CFD result. This is partially due to the rough inner surface of the present micro nozzle

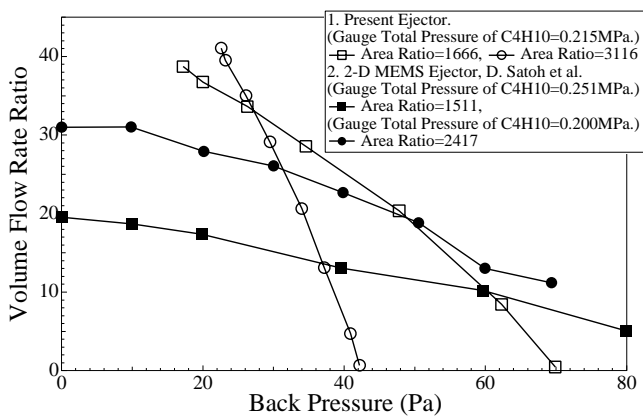


Figure 10. Comparison between present ejector and a 2-D MEMS ejector [3]. (Butane)

as shown in Fig. 3.

If compared with the 2-D MEMS ejector [3] having area ratio of 1511 and 2417, the present ejector produces higher volume flow rate ratio for small back pressure as shown in Fig. 10, although the ratio is more rapidly decreased with the increase of back pressure. It is conjectured that the volume flow rate ratio for higher back pressure can be larger if the secondary nozzle of present ejector has a diffuser for pressure recovery.

It has been found in the present simulation that the butane concentration is almost uniformly distributed in the radial direction at the exit of a 5 mm long secondary nozzle, when the gauge total pressure is 0.1 MPa. The tube length required for complete mixing becomes larger with increasing gauge total pressure (not shown).

To explore the possibility of applying other kinds of fuels having relatively lower molecular weight and higher vapor pressure as the primary flow source, a preliminary test with air has been conducted. A maximum volume flow rate ratio of 35 has been achieved for an ejector having area ratio of 3116 when back pressure is 12.2 Pa. The value is reasonably lower than butane due to its lower molecular weight. Theoretically, butane has the capability of producing a maximum volume flow rate ratio 1.4 times larger.

Figure 11 compares experimental data of the present ejector with 2-D MEMS ejectors [2] for air as the primary fluid. When the gauge total pressure is set as 0.2 MPa, the present ejector yields larger volume flow rate rate for small back pressure, although the ratio is more rapidly decreased. The data for gauge total pressure of air at 0.4 MPa indicates the potential of using low-molecular-weight gases for large back pressure conditions, because they can provide larger

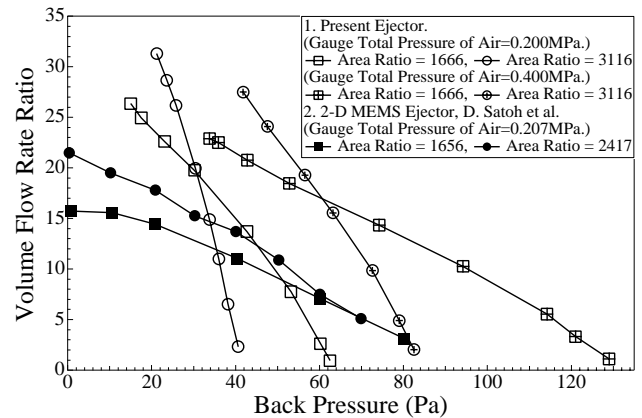


Figure 11. Comparison between present ejector and a 2-D MEMS ejector [2]. (Air)

volume flow rate ratio for large back pressure.

### 3 CONCLUSION

We have developed a micro ejector having an axisymmetric convergent-divergent nozzle to supply fuel-air mixture for a catalytic ceramic combustor. A volume flow rate ratio as large as 43 is achieved for a back pressure of 11.6 Pa, but it rapidly decreases with increasing the back pressure. The volume flow rate ratio obtained in the present experiment is much smaller than the CFD results. This is partially because the nozzle wall surface is somewhat rough.

### ACKNOWLEDGEMENT

This work is supported by the New Energy and Industrial Technology Development Organization (NEDO) of Japan.

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